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BCFD Anisotropic Adaptive Results Sonic Boom Prediction Workshop III

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Outline

- Tools and Computing Platform
- Summary of Cases
- Results: Biconvex 9x7 Shock-Plume Interaction Model
- Results: C608 Low Boom Flight Demonstration Model

Flow Solver/Computing Platform

BCFD Flow Solver (Boeing CFD)

- > 2nd order cell centered finite volume discretization (HLLE++)
- TVD Barth Limiter
- > All cases run with RANS SA turbulence model

EPIC (Edge Primitive Insertion and Collapse) Adaptation

- Coarsen/refines surface and volume mesh to match target metric field
- Sizing metric based on solution error estimate + geometric constraints

Computing Platform

- MPI parallel (BCFD), MPI+openMP parallel (EPIC)
- ▶ Bi-convex 10 nodes, 32 AMD EPYC 7301 CPUs, 96GB memory
- C608 400 cores, UV 3000 with Intel XEON E5-4650 v2 2.4 Ghz CPUs, 1.5TB memory

Summary of Cases/Grids

- Fixed grids provided by workshop (c608 visc-mixed)
- Adapted grids generated with EPIC (prism/tet element)
 - ✓ Very coarse initial grids, < 1M cells, y+ ~ 500
 - ✓ GO Goal Oriented output error (adjoint-based pressure error on all signature lines)
 - ✓ MS Multi-scale error (minimize Mach Hessian interpolation error)



Geometry Processing

- Started with provided STEP files for all cases
- Processed models in CADfix
 - Convert all surfaces to Nurbs
 - Export IGES plus topology file
- Modifications Performed
 - > Axie
 - Converted model from mm to meters
 - ✓ Split model at y=0 plane
 - Removed half body
 - > Biconvex
 - Split horizontal tail at leading edge line
 - ✓ Split nose cone



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Adapted Mesh + Interpolated Solution

- Iterative application of solution/adaptation at specified sequence of target cell sizes (complexity)
- This layer of BL elements inserted near wall



y+ = 1 normal spacing
~ y+ = 800 total thickness of prism layers

Adaptive Grids

- Coarse initial grids created for each configuration
 - Anisotropic triangular surface mesh (automatically refined to surface curvature)
 - Wall normal spacing y+ ~ 500
 - Prism/Tet elements (600K axie, 265K biconvex, 842K c608)

outflow boundary

Adapted Using 2 Error Estimates

Multiscale Error Estimate

- Minimize Lp interpolation error of Mach Hessian
- Targets adaptation to reduce error globally

Goal Oriented Error Estimate

- Minimize error in gauge pressure at signature lines
- Only adapts mesh where it impacts solution on signature lines





lower domain boundary

Biconvex Configuration – Adapted Grids



Sample Output Functional Grid Convergence – Biconvex



X-Force History

1.02

Sample BCFD Solution Convergence Biconvex 128M Cell - Goal Oriented Adapted Grid

- Solution convergence determined from residuals, forces, and sum of gauge pressure
- Initial solution interpolated from previous grid
- Solution convergence comparable with other BCFD adaptive cases







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Adaptive Grid Convergence – Biconvex Case (1M Cells)



Adaptive Grid Convergence – Biconvex Case (2M Cells)



Adaptive Grid Convergence – Biconvex Case (4M Cells)



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Adaptive Grid Convergence – Biconvex Case (8M Cells)



Adaptive Grid Convergence – Biconvex Case (16M Cells)



Adaptive Grid Convergence – Biconvex Case (32M Cells)



Adaptive Grid Convergence – Biconvex Case (64M Cells)



Adaptive Grid Convergence – Biconvex Case (128M Cells)



Goal Oriented Error Estimate



Multiscale Error Estimate



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Mach Contour Comparison - Biconvex Nozzle Plume Goal Oriented and Multiscale 128M cell Adapted Grids



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Adaptive Grid Convergence C608 Demonstrator



- Majority of signature captured at 64M cells
 Second nose expansion and nozzle wake
 - interaction slow to converge



- Majority of signature captured at 64M cells
- Additional peaks in signature aft of nozzle

Comparison of c608 Near Field Signature Finest Fixed/Adapted Grids













Goal Oriented





Conclusions

Grid Convergence

- Adaptivity improves mesh convergence over fixed grid (c608 case)
- GO convergences faster than MS for bi-convex and c608 cases
- GO advantage less prominent for c608 due to higher geometry fidelity and additional coverage of signature lines

Nozzle Plume Interaction

- Bi-convex case: secondary compression resolved in GO results
 > better resolution of nozzle plume and h-tail reflected shock interaction
- C608 demonstrator: significant differences in signature aft of nozzle
 - additional signature peaks in MS and fixed grid results

Analysis Challenges

- Tecplot macro failed to run on 401M cell fixed grid
- Primal solution failed to converge for some c608 256M cell grids